



# Interferometric Imaging and Stellar Interferometry at JPL

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## Why Interferometry?



#### 1: Imaging resolution

- $\lambda$ **B** for an interferometer vs.  $\lambda$ **D** for a telescope
- B, separation of apertures, can cost-effectively be made very large

#### 2: Astrometric accuracy

- Interferometers have a simple geometry which can be accurately monitored to minimize systematic errors
- Interferometers use starlight efficiently

#### 3: High dynamic range capability

- Interferometers have the ability to "null starlight" with extreme precision
- Allows detection of planets or other dim objects orbiting a star





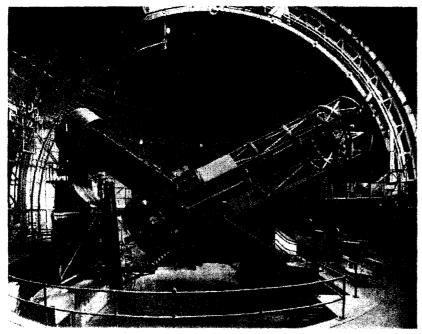


Figure 13.5 The 100 inch (2.5 m) Hooker reflector on Mount Wilson, completed in 1917. (Courtesy The Observatories of the Carnegie Institution of Washington.)



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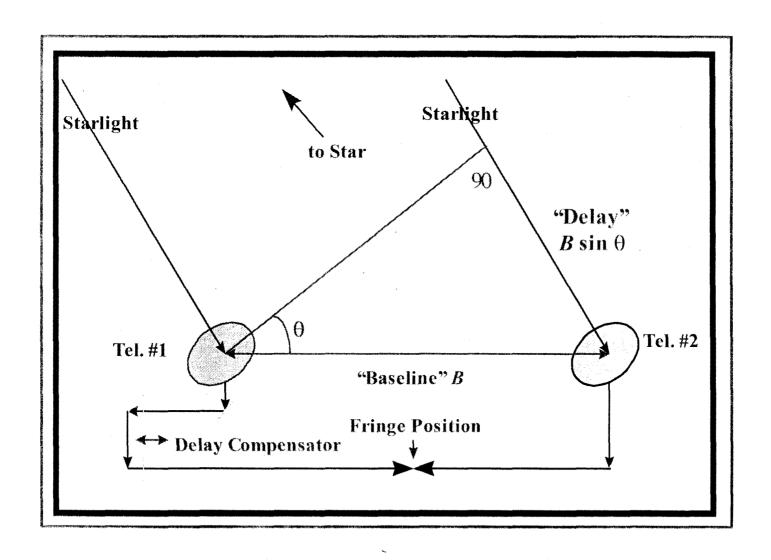
PEET

Fig. 1.—Diagram of optical path of interferometer pencils.  $M_1$ ,  $M_2$ ,  $M_3$ ,  $M_4$ , mirrors; a, 100-inch paraboloid; b, convex mirror; c, coudé flat; d, focus.



# Simple Long-Baseline Stellar Interferometer JPL

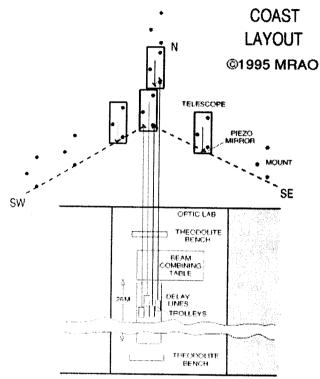




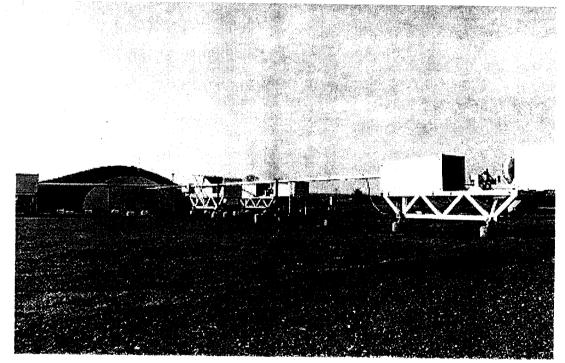


# Optical Imaging through Stellar Interferometry

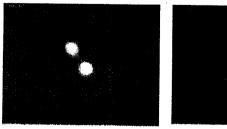




#### **Cambridge Optical Aperture Synthesis Telescope (COAST)**



**Optical Synthesis Images of Capella** 



13 Sept. 1995

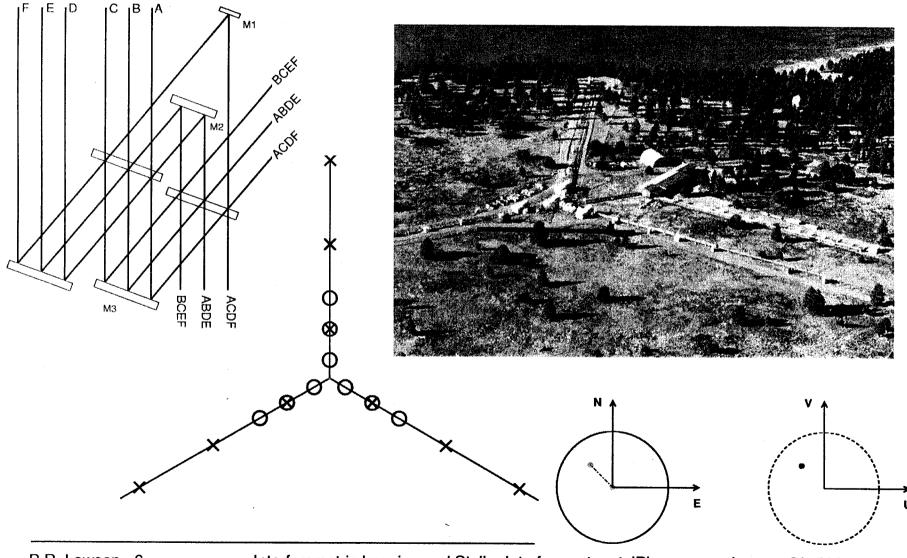


28 Sept. 1995



# Beam Combination for Synthesis Imaging

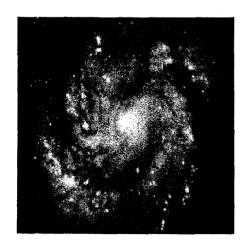




Interferometric Imaging and Stellar Interferometry at JPL

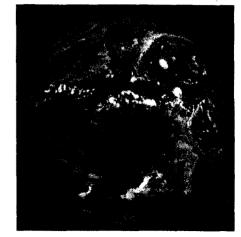
January 26, 2001.

#### NASA's Astronomical Search for Origins



To understand how galaxies formed in the early universe and to determine the role of galaxies in the appearance of stars, planetary systems and life.

To understand how stars and planetary systems form and to determine whether life-sustaining planets exist around other stars.





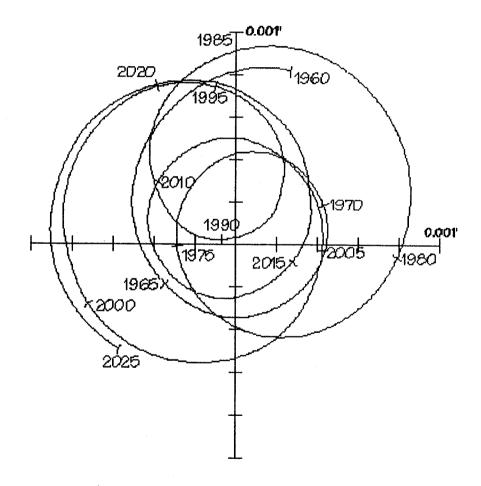
To understand how life originated on Earth and to determine if it began and may still exist elsewhere as well.



#### Astrometric Search for Planets



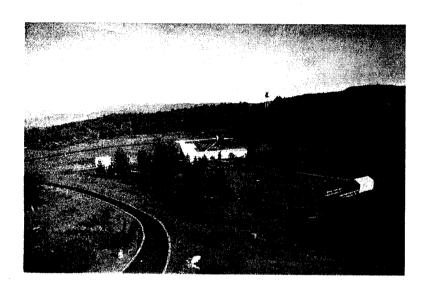
- Astrometry measures positional wobble due to planets
  - More sensitive than RV
  - no sin(i) ambiguity
- Interferometry will enable measurements at the micro-arcsecond level
  - Earth at 10 pc produces 0.3 μas amplitude
- Result of new observing systems will be a census of planets down to a few M<sub>earth</sub> over the next 10-20 years

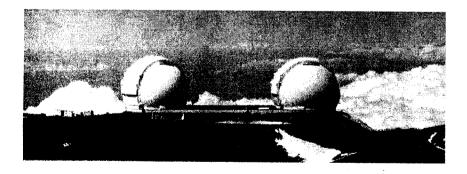


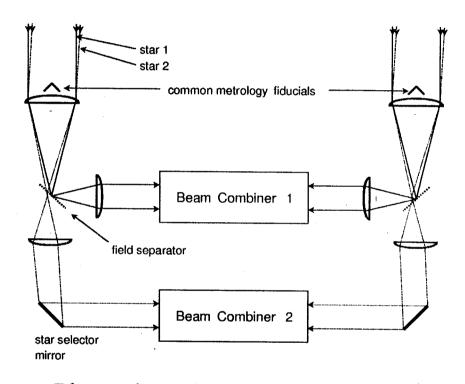


#### Narrow Angle Astrometry









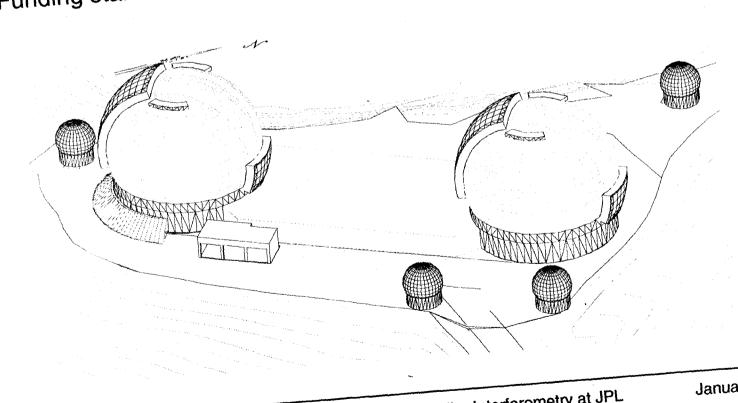
- Planet detection: "Jupiter" at 10 pc produces a 1 mas peak-to-peak star wobble.
- •20 µas astrometry possible from the ground over narrow angles.



# Keck Interferometer



- Interferometry with the two 10-m Keck telescopes and four 1.8-m outrigger telescopes
- Funding from NASA as part of the Origins Program
- Joint development between JPL and CARA
- Funding started in Oct 1997 (FY98); CDR: Aug '98





### Astrometric Detection of Exoplanets

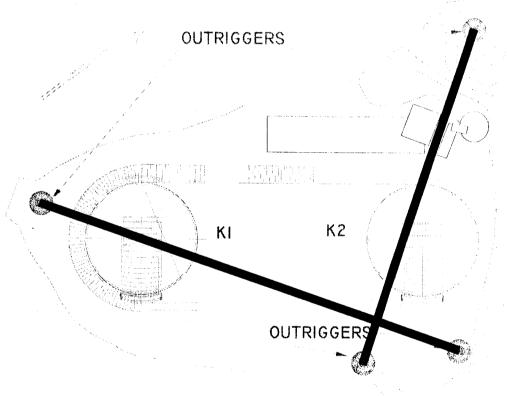


#### Science objective

- Survey 100's of nearby stars for planets to Uranus mass
- Uses outrigger telescopes for long-term survey

#### Approach

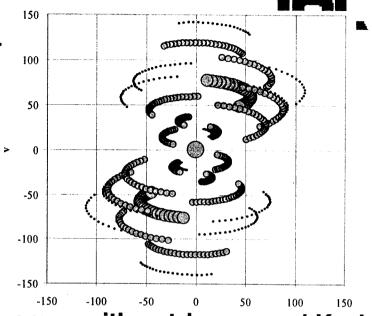
- High-accuracy narrow-angle astrometry
- Configuration
  - 4 1.8-m outriggers
  - Orthogonal >100m baselines
  - Dual-star feeds
  - End-to-end laser metrology
  - 30 μas per hour accuracy for differentia astrometry
    - Uses K17 isoplanatic ref.





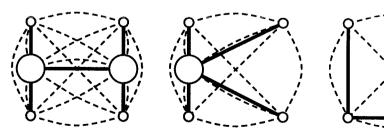
#### **Imaging**

- Imaging with 6-element array
- Good (u,v)-plane coverage
- 9 of 15 baselines include a 10-m telescope
  - Background-limited sensitivity equivalent to two 4.4-m's
  - Other imaging options using OTs with 1 or 0 Kecks
- Use cophasing to increase sensitivity
- Ultimate sensitivity
  - K-K fringe tracking: K14.1
  - OT-OT fringe tracking: K10.8
  - Cophased imaging
    - K19 for SNR >10 in 1000 s on a Keck-OT baseline
    - Requires K11.6 isoplanatic ref



u,v coverage with outriggers and Kecks (width of line indicates baseline).

#### Science and cophasing baselines

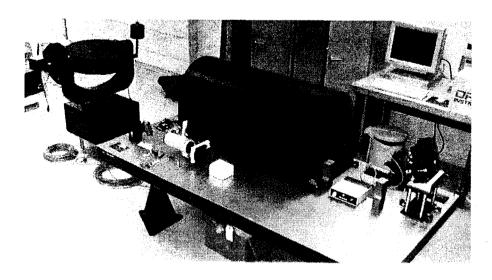


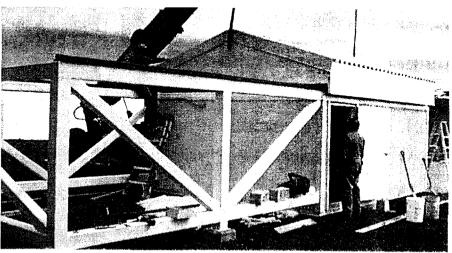
\_\_\_ cophasing science



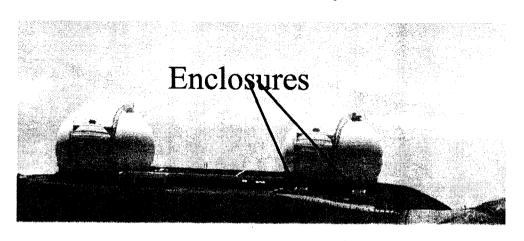
#### **Siderostats**







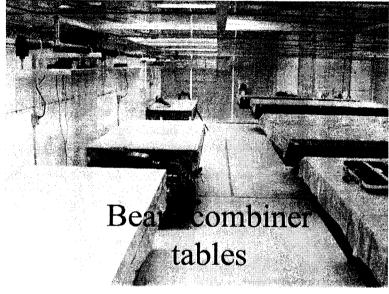
- Siderostats installed & operational on summit
- Beam transport mirrors aligned & star acquired in basement



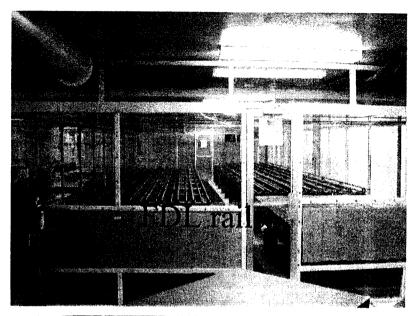


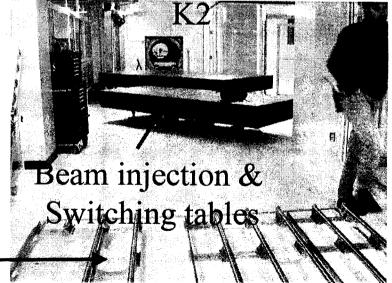
#### **Basement Infrastructure Installation**







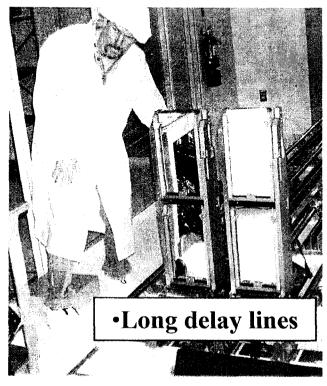


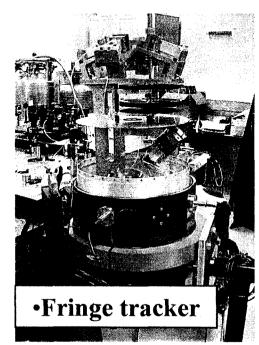




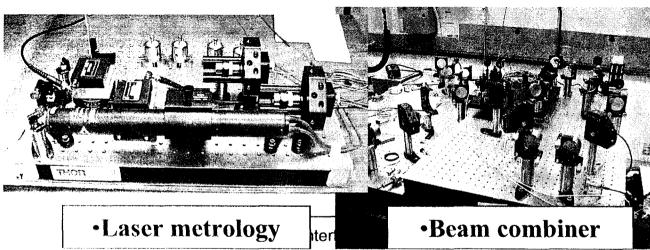
#### Hardware

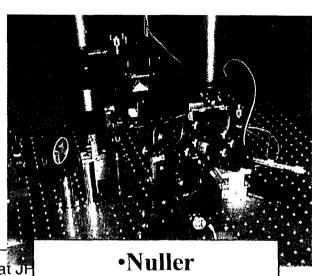


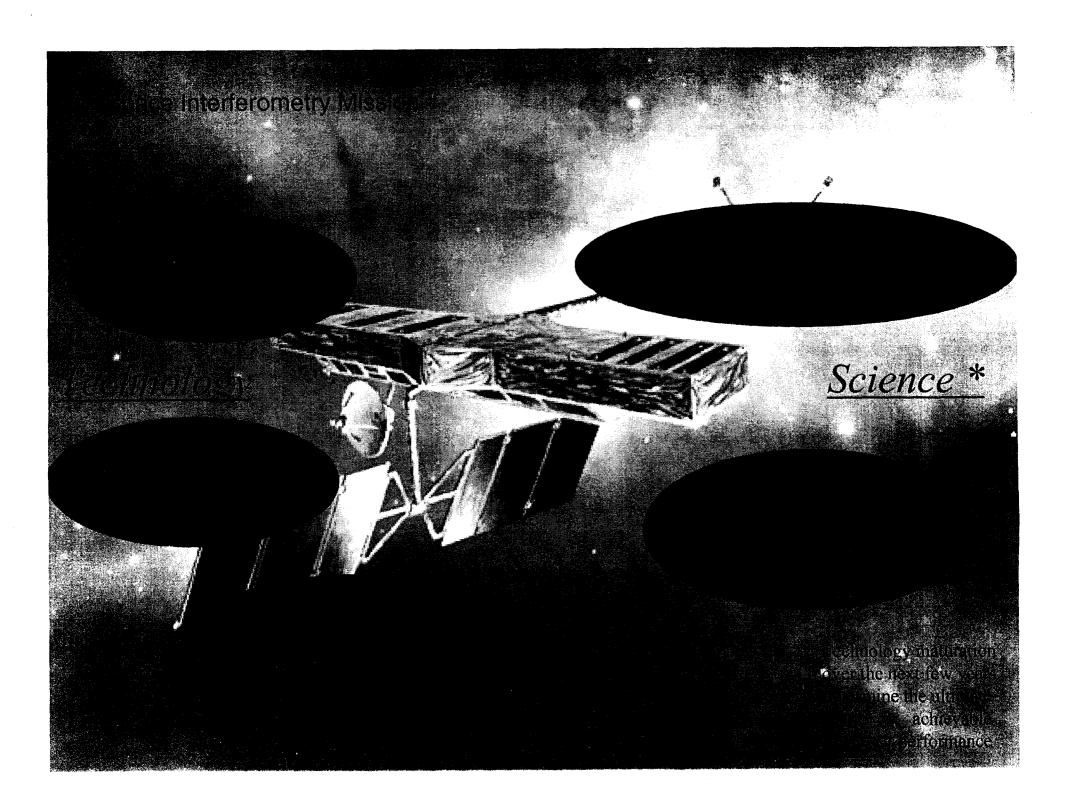








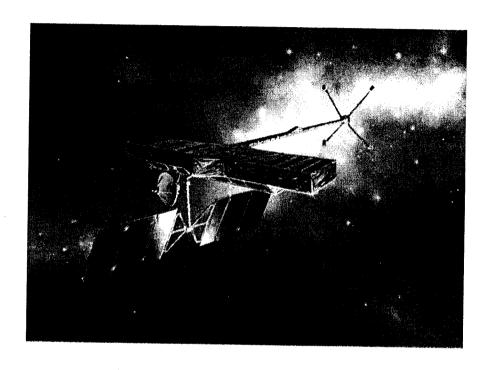






# Space Interferometry Mission (SIM)



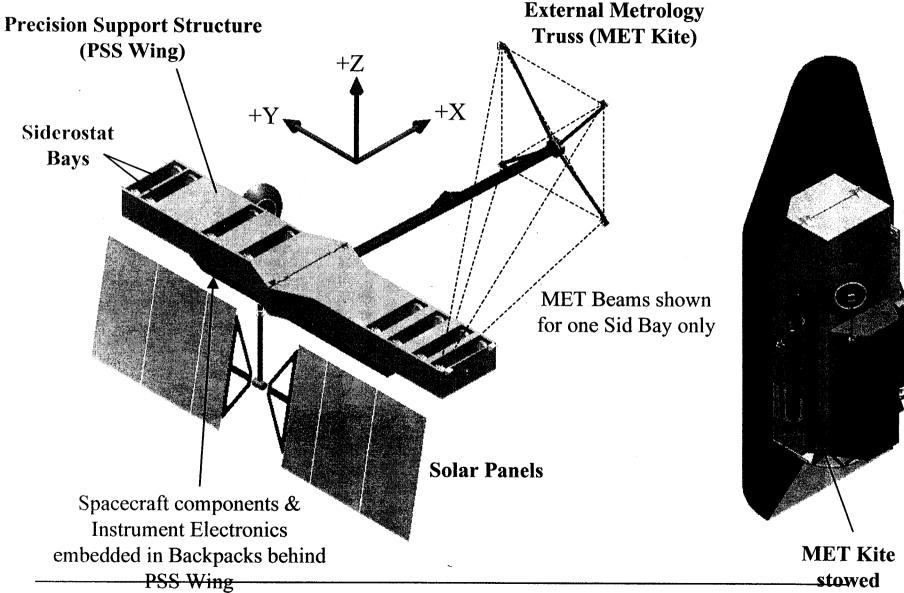


Wide angle astrometric accuracy  $4 \mu as (5 yr)$ Narrow angle astrometic accuracy  $1 \mu as (1 hr)$ Instrument field of regard 15 degreesInstrument throughput (V=16 star)  $5.8 \mu as (10 min)$ Synthesis Imaging Resolution 10 masNulling Depth  $10^{-4}$ 



### SIM Configuration

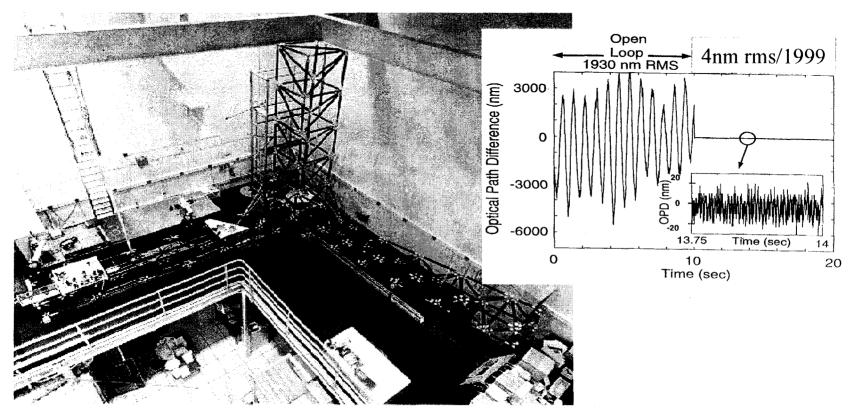






#### Nanometer Control Testbed



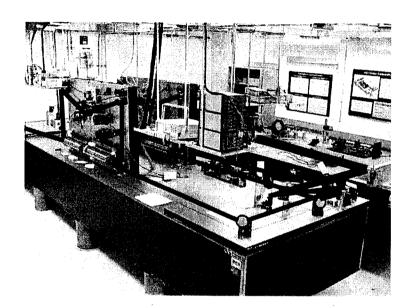


Flexible truss ~5 hz resonance Simulated spacecraft disturbance (reaction wheels) Active isolation of disturbance & active optical loop (using laser interferometry as the sensor)



## SIM System Testbed (STB-3)





#### Now: 3 baselines on optical table

- Three interferometers functioning on an optical table
- Completed detailed design of SIMscale flexible structure to be built and installed by end-2000



- Begin nanometer active control experiments on flexible structure
- Three baselines, full scale

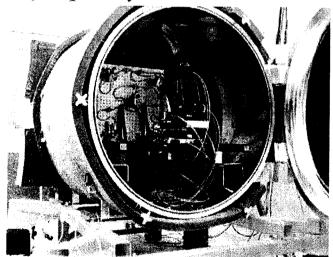


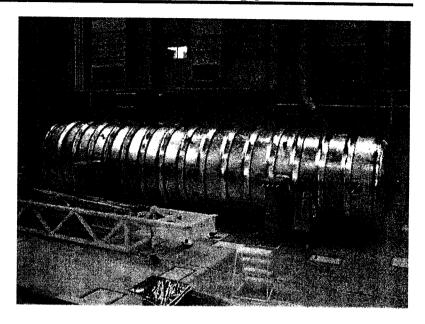
## Picometer measurement technology



#### Component technologies

- superprecise optical elements
- picometer laser gauges
- •frequency stabilized lasers





Microarcsec Metrology Testbed
• fully functional interferometer
tested in vacuum at picometer
levels

• verify testing procedure for flight hardware

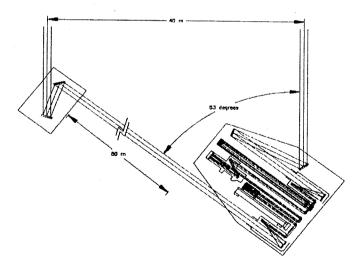


# Multiple Spacecraft Interferometry ST-3 Technology Mission



- The New Millennium Program ST-3
   Mission will provide validation of key
   enabling technologies for TPF
  - Separated S/C interferometry
  - Precision formation flying
  - Real-time optical control of a separated S/C interferometer
  - Angular and linear metrology
  - Inertial referencing for phasing and guiding
  - Separated S/C interferometer I&T techniques

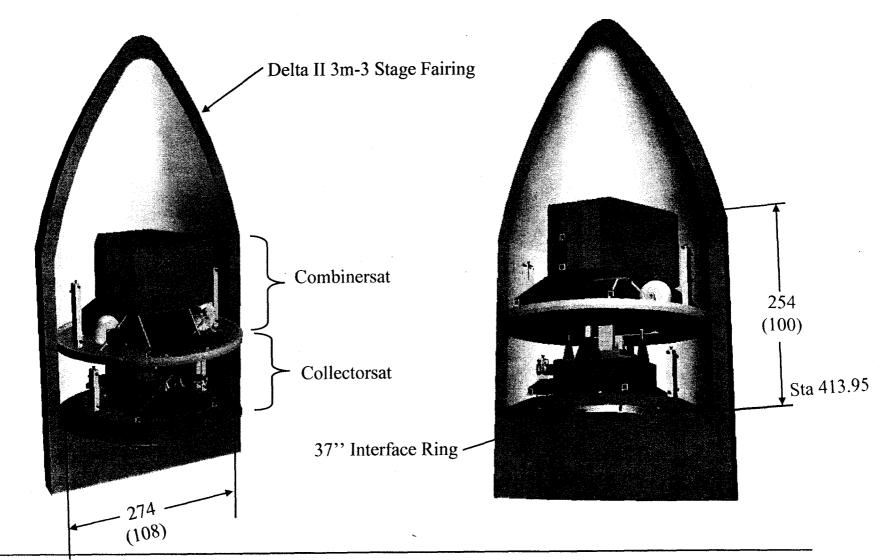






# ST3 Launch Stack Configuration

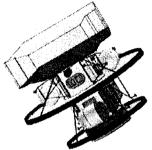


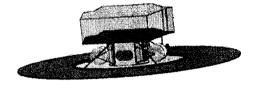




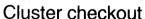
#### **Mission Timeline**



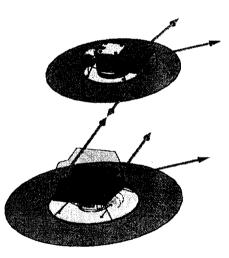


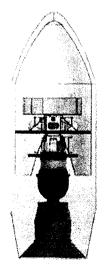


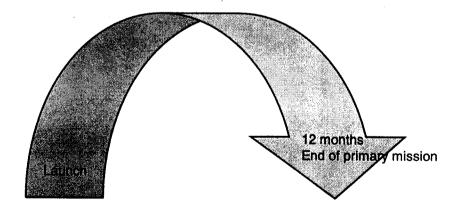
S/C separate and perform formation flying checkout & experiments Combiner mode interferometry checkout, experiment & observations











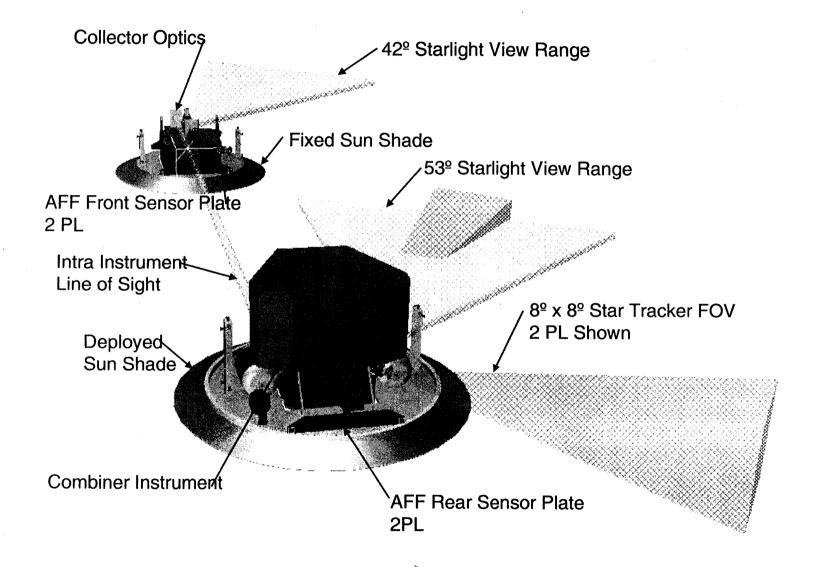
Delta 2925 Launch - September 2005 Heliocentric Earth trailing orbit

Separated spacecraft interferometry checkout experiments & observations Formation flying experiments



## Flight System On-Orbit Configuration

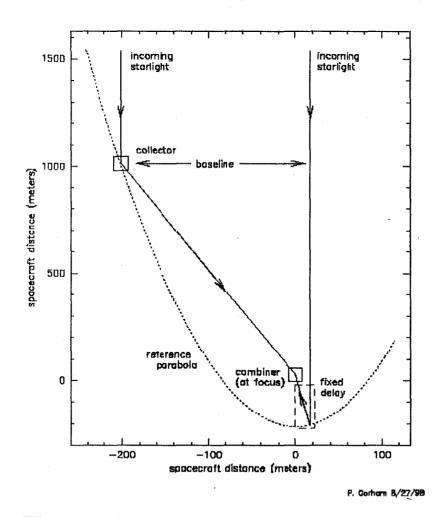




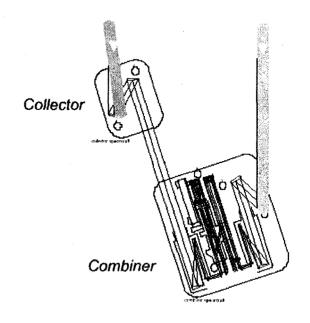


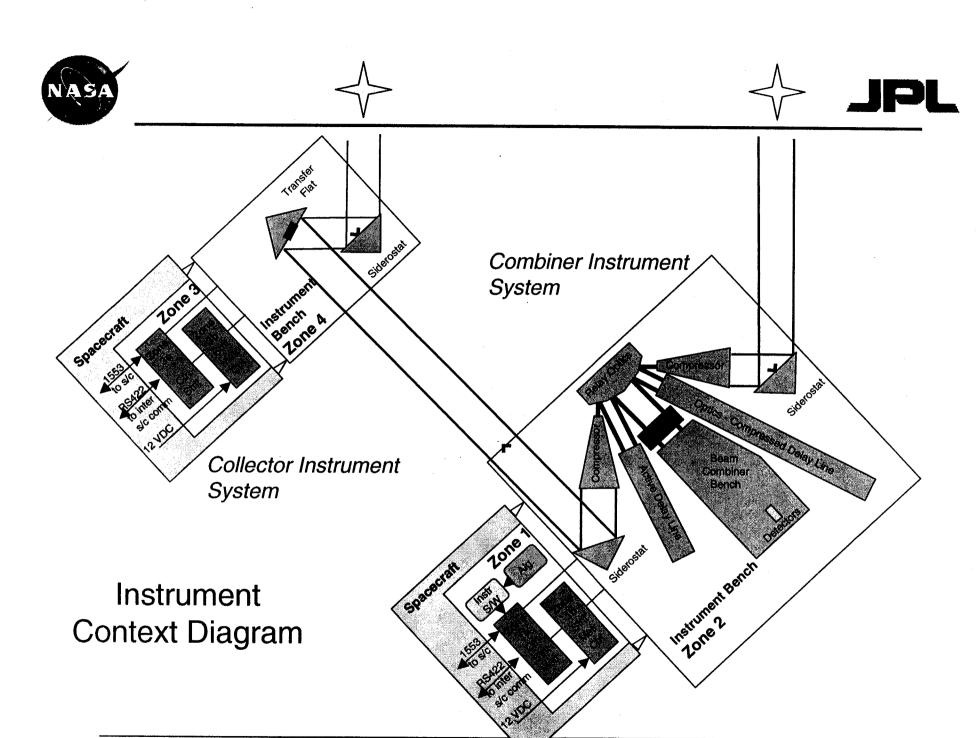


# Two Spacecraft Interferometer Concept



- S/C separation from 50 m to 1 km
  - Observation baselines of 40 to 200 m
  - 8th magnitude stellar targets
- · Parabola is locus of constant delay
- Combiner contains 20 m fixed delay line
- Combiner can operate as a 1 m monolithic interferometer
  - No collector, bypass fixed delay

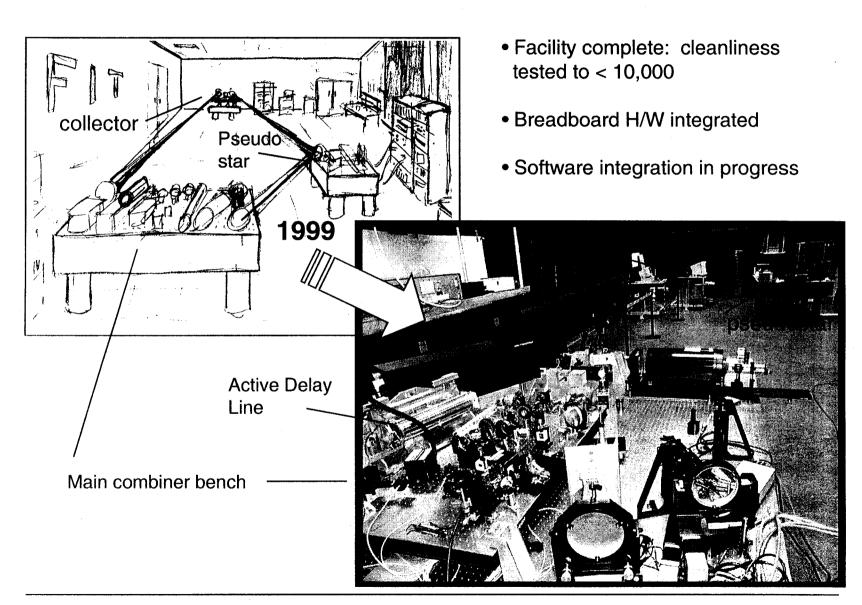






#### Formation Interferometer Testbed

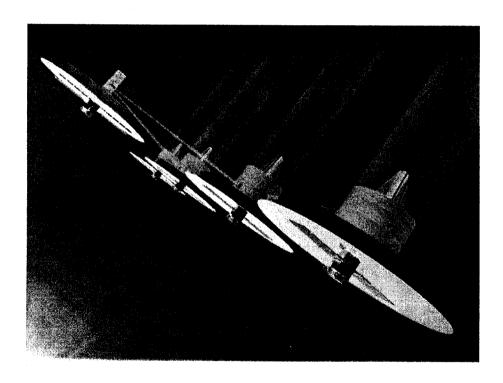






### Terrestrial Planet Finder (TPF)





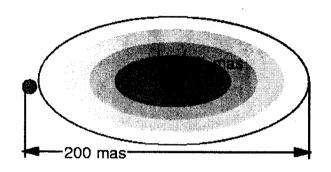
- Direct detection of Earth-like planets around nearby stars
- Approach
  - Interferometric starlight nulling by ~ 10<sup>6</sup> to detect 10 um infrared light from the planet
  - ~10 hrs of observation to detect an Earth-like planet at10 pc
  - 2-4 weeks to measure a low resolution spectra of the atmosphere, to identify H<sub>2</sub>O, CO2, O3

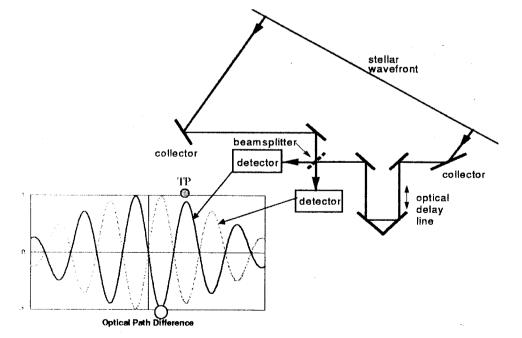
- Configuration
  - 4 large (~3m) collecting apertures
  - Cryogenically-cooled optics
  - Formation flying to provide baselines of 50 - 500 m

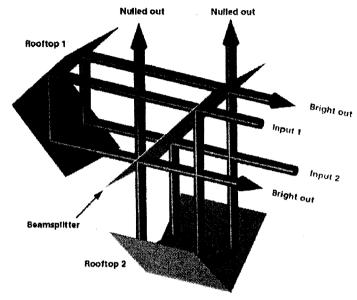


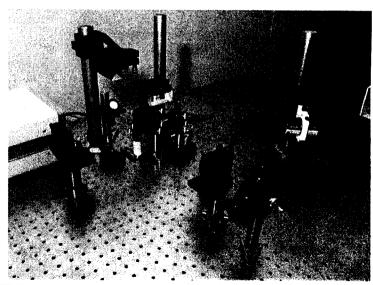


## **Nulling Interferometry**











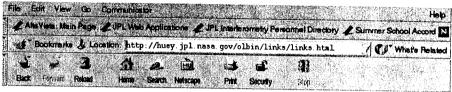
## Optical Long Baseline Interferometry Newsletter



http://huey.jpl.nasa.gov/olbin/

CL 99-2072

- Contains links to all existing and proposed optical/IR interferometer projects.
- News
- Papers and preprint information
- **Upcoming meetings**
- **Contact information**
- Translations of selected papers
- **List of PhD and Masters theses**
- Photographs and resources
- **Job listings**



#### JPL Interferometry Center of Excellence

Optical Long Baseline Interferometry News

#### Links

Home | News | Links | Meetings | Publications | 1998 Papers | 1999 Papers | 2000 Papers | Preprints | Software | Phonebook | Photos | Theses | Stereograms | Jobs | Weather | Log

#### LONG BASELINE STELLAR INTERFEROMETERS

#### Operational Ground-Based Interferometers

- CHARA Array Center for High Angular Resolution Astronomy Array (Mnt Wilson, California; GSU)
- COAST Cambridge Optical Aperture Synthesis Telescope (MRAO, UK).
- FLUOR Fiber Linked Unit for Optical Recombination (Obs. de Paris, France) GI2T/REGAIN – Grand Interféromètre à 2 Télescopes (Obs. de Calem, France).
- IOTA Infrared-Optical Telescope Array (Whipple Obs., Arizona). Photo.
- ISI Infrared Spatial Interferometer (Mount Wilson Obs., California).
- NPOI Navy Prototype Optical Interferometer (Lowell Obs., Arizona). Photos.
- PTI Palomar Testbed Interferometer (Mount Palomar, California).
- SUSI Sydney University Stellar Interferometer (Narrabri, Australia).

The following interferometers have ceased operations - the indicated dates are approximately the dates that the interferometers were operational: 20ft (1920-1931), 50ft (1931-1938), Intensity Interferometer (1964–1976), I2T (1974–1987), Mark I (1979), Mark II (1982–1984), 11.4m prototype (1985–1988), SOIR DETE (1979-1993), IRMA (1990-1992), Mark III (1986-1993), IZT/CHARON (1993-1996), and MIRA-I

A Timeline of Stellar Interferometry from 1955 to the present shows the development of these interferometers.

**Ground-Based Interferometers Under Construction** 



### Acknowledgments



Space Interferometry Mission: Michelson Fellowship Program

Terrestrial Planet Finder Preproject Office

JPL Technology and Applications Program Directorate

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